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MEMORANDUM FOR PRS (In-House Publication)

FROM: PROI (STINFO)

27 June 2002

SUBJECT: Authorization for Release of Technical Information, Control Number: AFRL-PR-ED-VG-2002-166 C.T. Liu (PRSM), "Investigating the Constraint Effect in a Particulate Composite Material" (viewgraphs)

ASME Pressure Vessel Technology Meeting (Vancouver, Canada, 7-9 August 2002) (<u>Deadline: 30 July 2002</u>) (Statement A)

b.) military/national critical technology, c.) ed.) appropriateness for release to a foreign national comments:	ation, and e.) technical sensitivity and/or economic sensitivity.
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and/or b) possible higher headquarters revie Comments:	
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b) appropriateness of references, if applicab	TNFO for: a.) changes if approved as amended, le; and c.) format and completion of meeting clearance form if required
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	PHILIP A. KESSEL Date

Technical Advisor

Space and Missile Propulsion Division



C. T. Liu AFRL/PRSM 10 E. Saturn Blvd. Edwards AFB CA U.S. A. 93524-7680,



Objectives



- Intensity Factor, K_{li}, for the Onset of Crack Growth in a Investigate the Constraint Effect on the Critical Stress Particulate Composite Material
- Specimen Thickness: 0.2 in., 0.5 in., 1.0 in., 1.5 in.
- Initial Crack Length: 0.2 in., 0.3 in., 0.4 in.



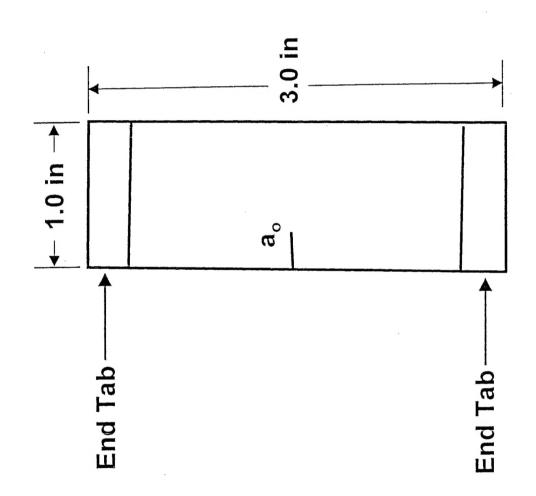


- Experimental findings indicate that, on the first thickness and initial crack length for the cases approximation, K_{II} is independent of specimen considered in this study.
- Due to the development of damage at the crack tip, the constraint is minimized at the crack tip and the Poisson's effect is negligible.
- investigation, the plane-strain fracture toughness does For the particulate composite material under not exist



Specimen Geometry

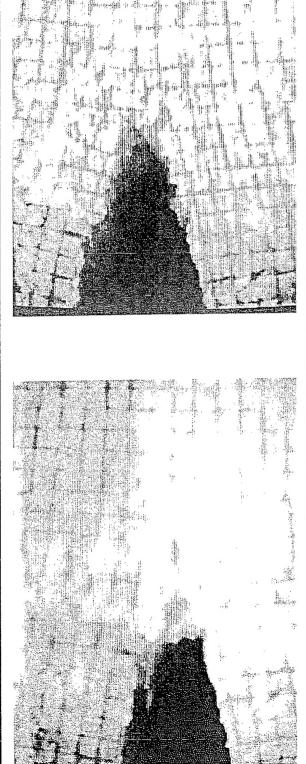


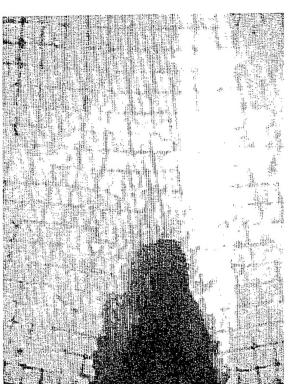




(crosshead speed = 0.508 mm/min)





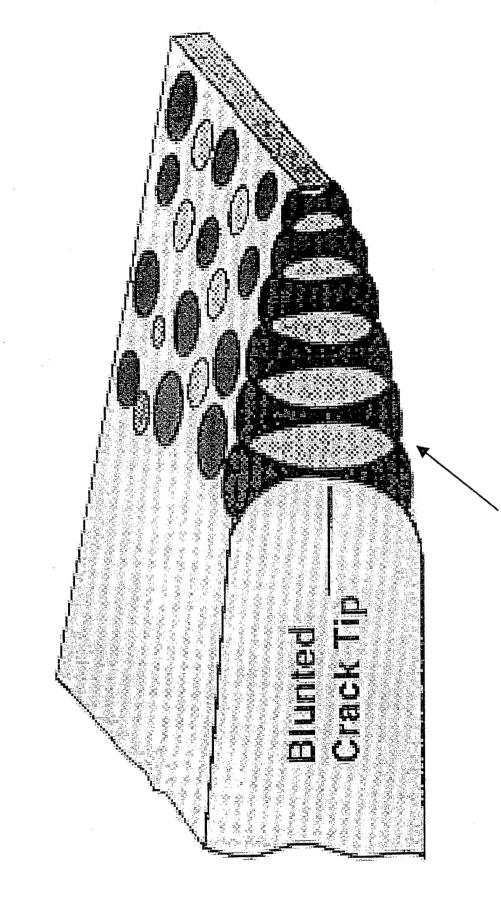






Crack I to Damage Mode

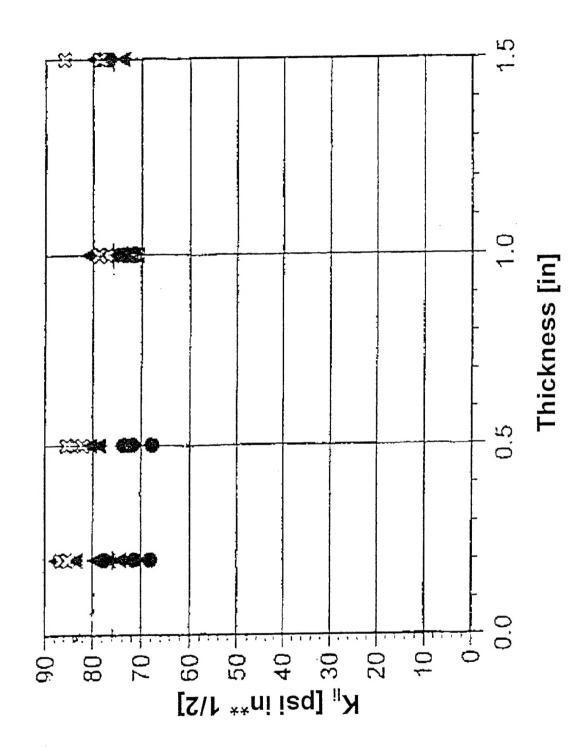




Highly Damaged Zone



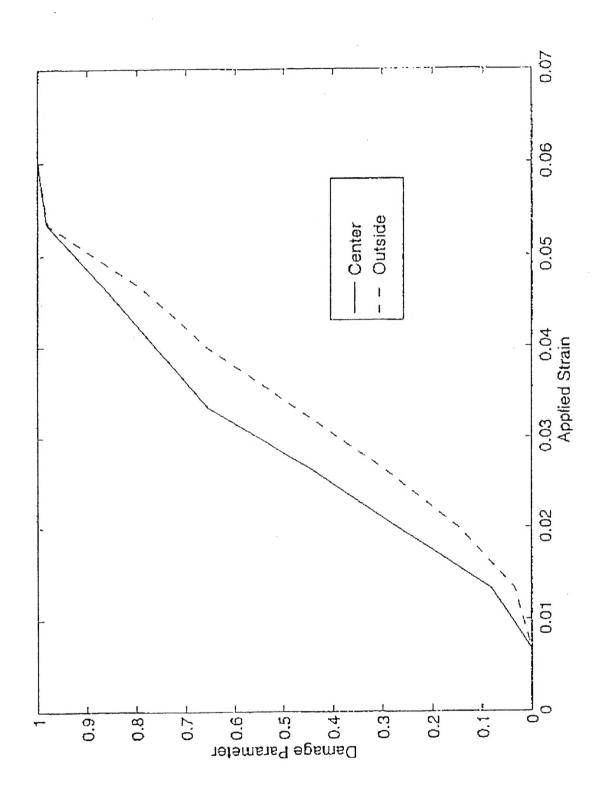






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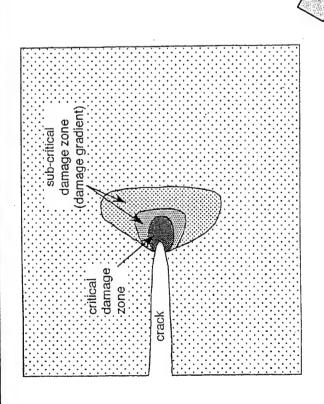


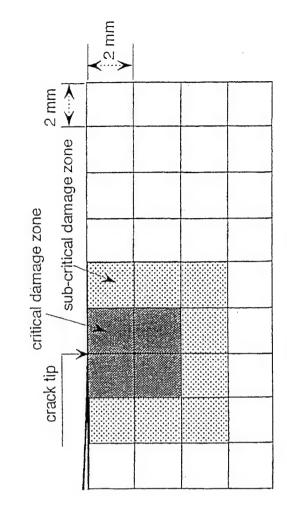




Finite Element Model







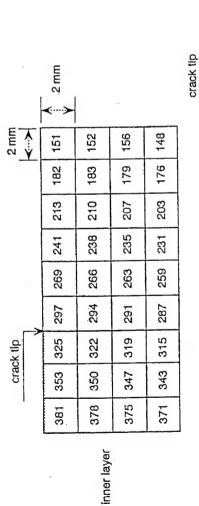
FEM mesh at crack tip





Finite Element Models of the Three Layers of the Analyzed Specimen





	154	153	157	149	
	185	184	180	177	
	214	211	208	204	
	242	239	236	232	
	270	267	264	260	
	298	295	292	288	
	326	323	320	316	
	354	351	348	344	
•	382 35	379	376	372	
		middle layer			

159	155	158	150
187	186	181	178
215	212	209	205
243	240	237	233
271	268	265	261
299	296	293	289
327	324	321	317
355	352	349	345
383	380	377	373
	outer layer		

crack tip



Summary of Crack-damage Interaction Analysis

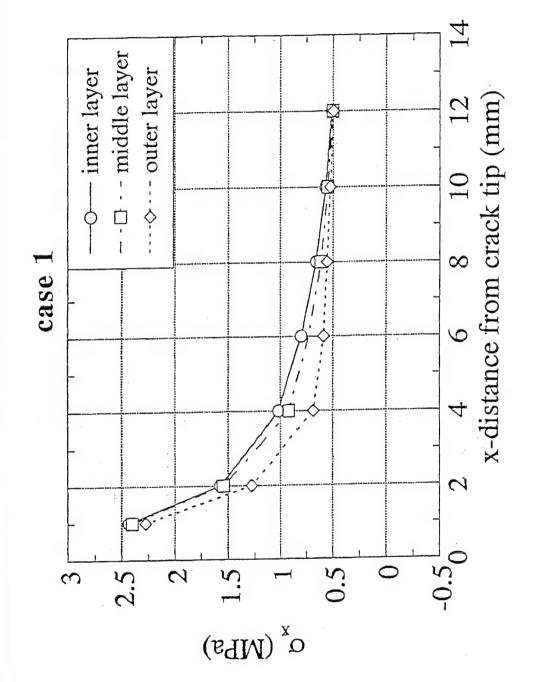
case	damage element	damage element	Poisson's ratio	inside layer	middle layer	outside layer
		modulus MPa (psi)		KI MPa-cm ^{0.5}	KI MPa-cm ^{0.5}	K _I MPa-cm ^{0.2}
thin*	none	0.414 (60)	0.4999	1.871	-	1
1	none	1	0.4999	1.931	1.903	1.802
7	325, 297	0.414	0.4999	0.422	2.246	1.871
3	325, 297 326, 298	0.414	0.4999	0.535	0.440	2.208
4	325, 297 326, 298 327, 299	0.414 (60)	0.4999	0.573	0.524	0.455
2	325, 297, 322, 294	0.414 (60)	0.4999	0.392	2.285	1.906
9	325, 297, 322, 294 326, 298, 323, 295 327, 299, 324, 296	0.414 (60)	0.4999	0.522	0.497	0.432
7	325, 297, 322, 294 326, 298, 323, 295 327, 299, 324, 296 353, 350, 347, 319 291, 263, 266, 269 354, 351, 349, 320 292, 264, 267, 270 355, 352, 349, 321 293, 265, 268, 271	0.414 (60) 0.828 (120)	0.4999	0.546	0.514	0.442
8	325, 297, 322, 294 326, 298, 323, 295 327, 299, 324, 296	0.414 (60)	0.1	0.336	0.339	0.342
,			000			A

* thin specimen, specimen thickness = 0.508 cm.



Stress Distributions in the Horizontal Direction

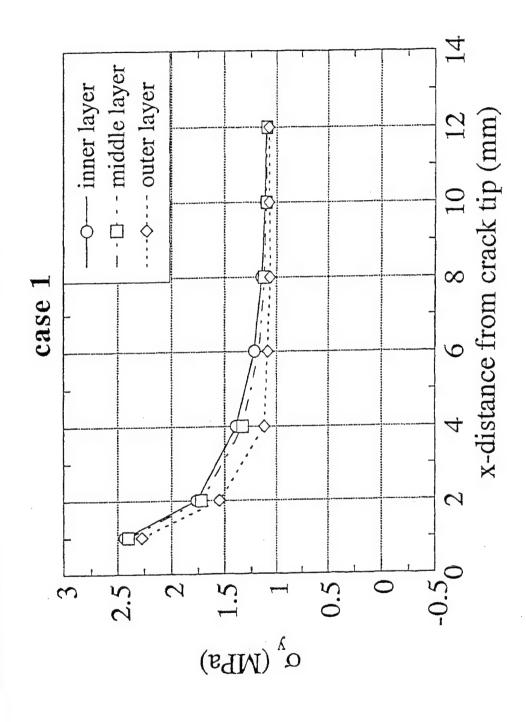






Stress Distributions in the Vertical Direction



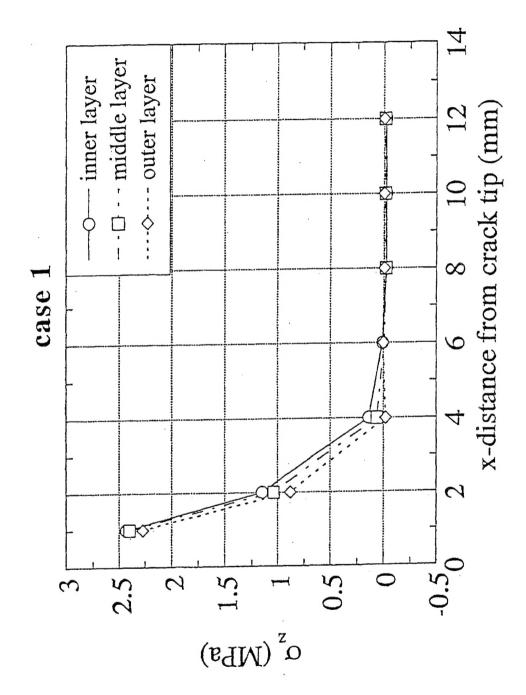






Stress Distributions in the Thickness Direction



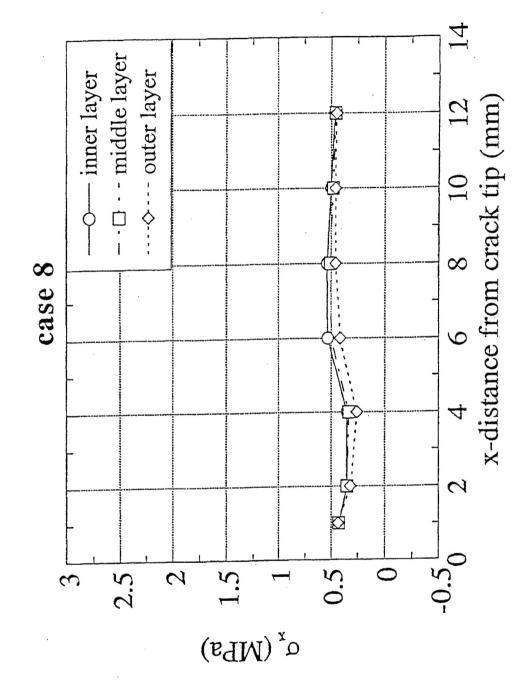






Stress Distributions in the Horizontal Direction

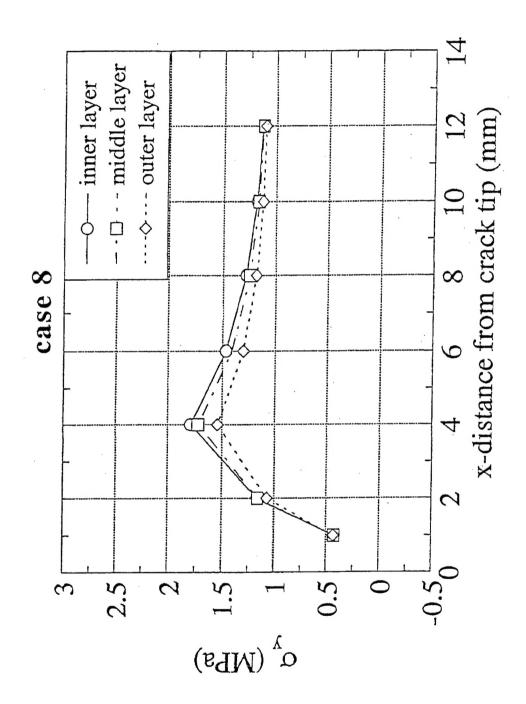




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Stress Distributions in the Vertical Direction







Stress Distributions in the Thickness Direction



